

# SAE AERO DESIGN PROJECT PROPULSION & P.I CONTROL SYSTEMS UNION COLLEGE FLIGHT CLUB

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# SAE Aero Design Team

- Society of Automotive Engineers
- Team of five, two EE's and three ME's
- Regular and Advanced Class Competitions



# Regular Class Competition Summary

- Design, build, and test an R/C aircraft
  - Carry max pay loads within 200 ft runway
  - Max combined L,W,H of 175 inches
  - Electric Propulsion System
  - Power limiter (1000W)
- Scoring
  - Bonus Points
  - Penalties Include
    - Power consumption >1000W
    - Bad take off/landing
    - Design modifications after submitting design Report

Round	R <sub>n</sub>	B <sub>n</sub>	T	FFS
1	10.70	0.00	5.00	
2	0.00	0.00	0.00	
3	16.40	4.00	3.00	
4	17.85	8.00	0.00	
5	18.02	19.00	0.00	
6	16.41	4.00	0.00	
<b>SUM=</b>	<b>79.38</b>	<b>19.00</b>	<b>8.00</b>	<b>90.38</b>

$$FFS = \sum_1^n R_n - \sum T + B_{n(max)}$$



# OPTIMIZING POWER CONSUMPTION OF SAE AERO AIRCRAFT

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By Ervin Meneses





# Motivation

Last years competitors!

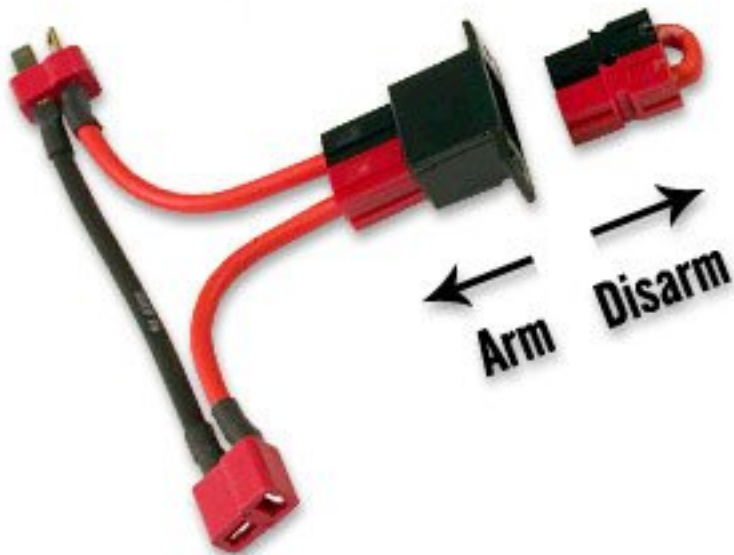
# Presentation Overview

- Design Requirements
- Goals
- Propulsion Systems
- Testing
- Future Work



# Design Requirements

- Single electric motor configuration
- Use of one COMMERCIAL Li-Po Battery
  - Min. Req. of 3000mAh @ 25C
- Use of 2015 Power Limiter
- Install Red Arming Plug



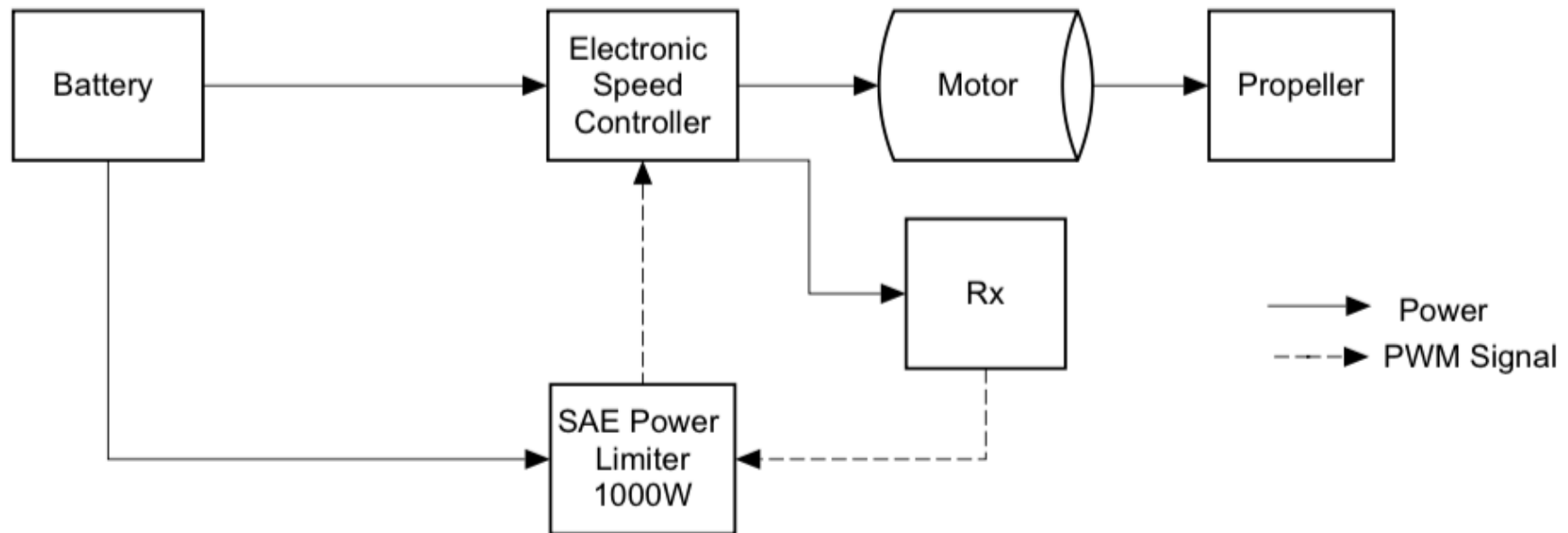
# Goals

- Competition
  - Provide the team with an aggressive electric propulsion system (EPS) that complies with SAE Aero Rules and provides more than 11 lbf of thrust
- Senior Project
  - Design and implement a P.I controller that will serve as a pre limiting device





# 2014 EPS



# Battery, Propeller, Motor, & ESC Selection

- Enough energy for flight
- Propeller with greatest thrust
- Same motor and ESC from last year



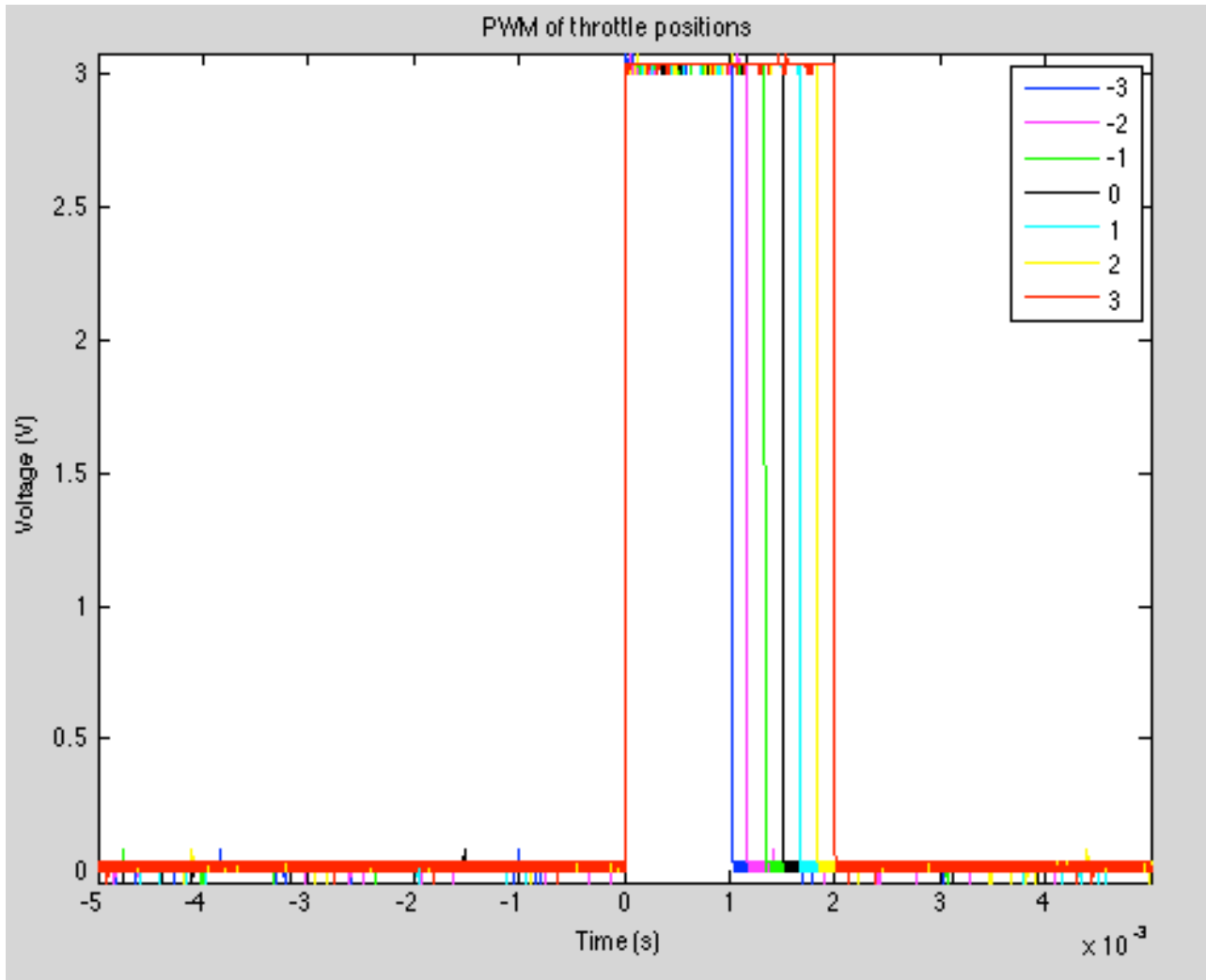
1lb 6oz



1lb 2oz



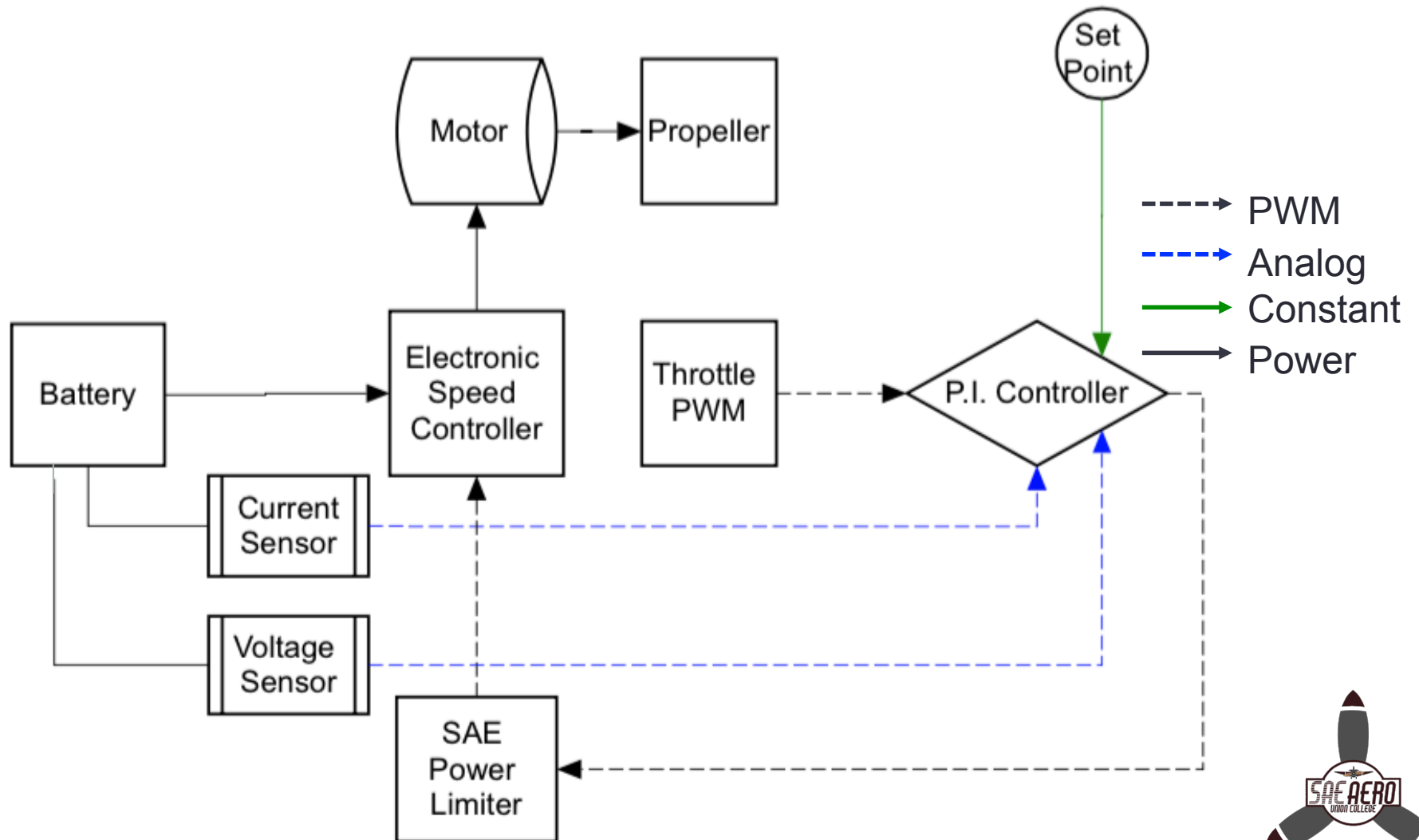
# Control Signal



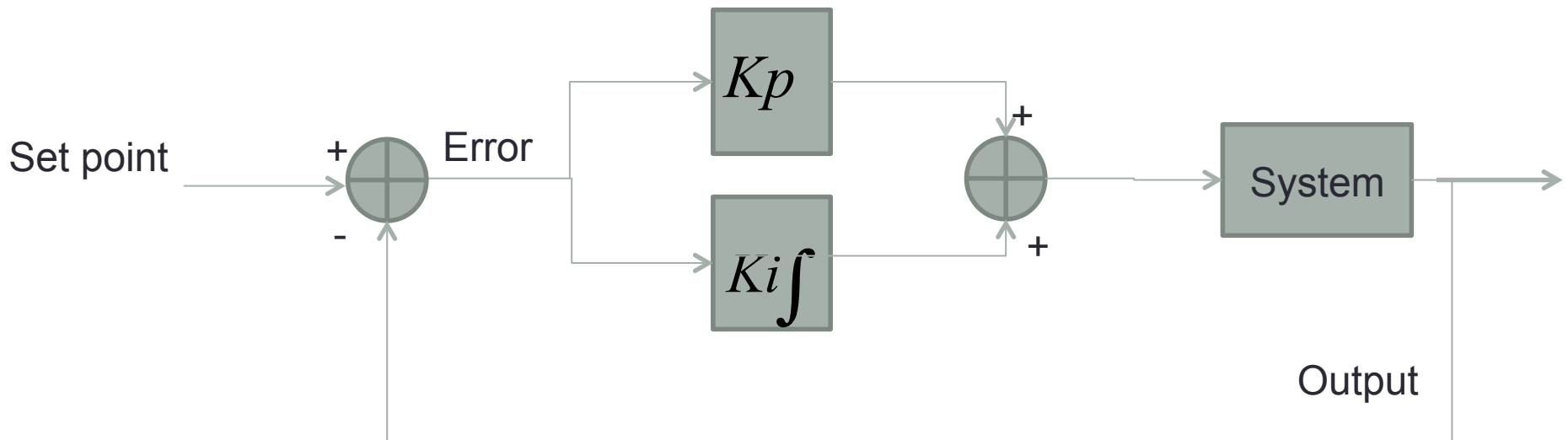
Throttle Position	PWM IN Rx(45 Hz)
-3	908 us
-2	1.052ms
-1	1.272ms
0	1.508ms
1	1.660ms
2	1.844ms
3	2.004ms



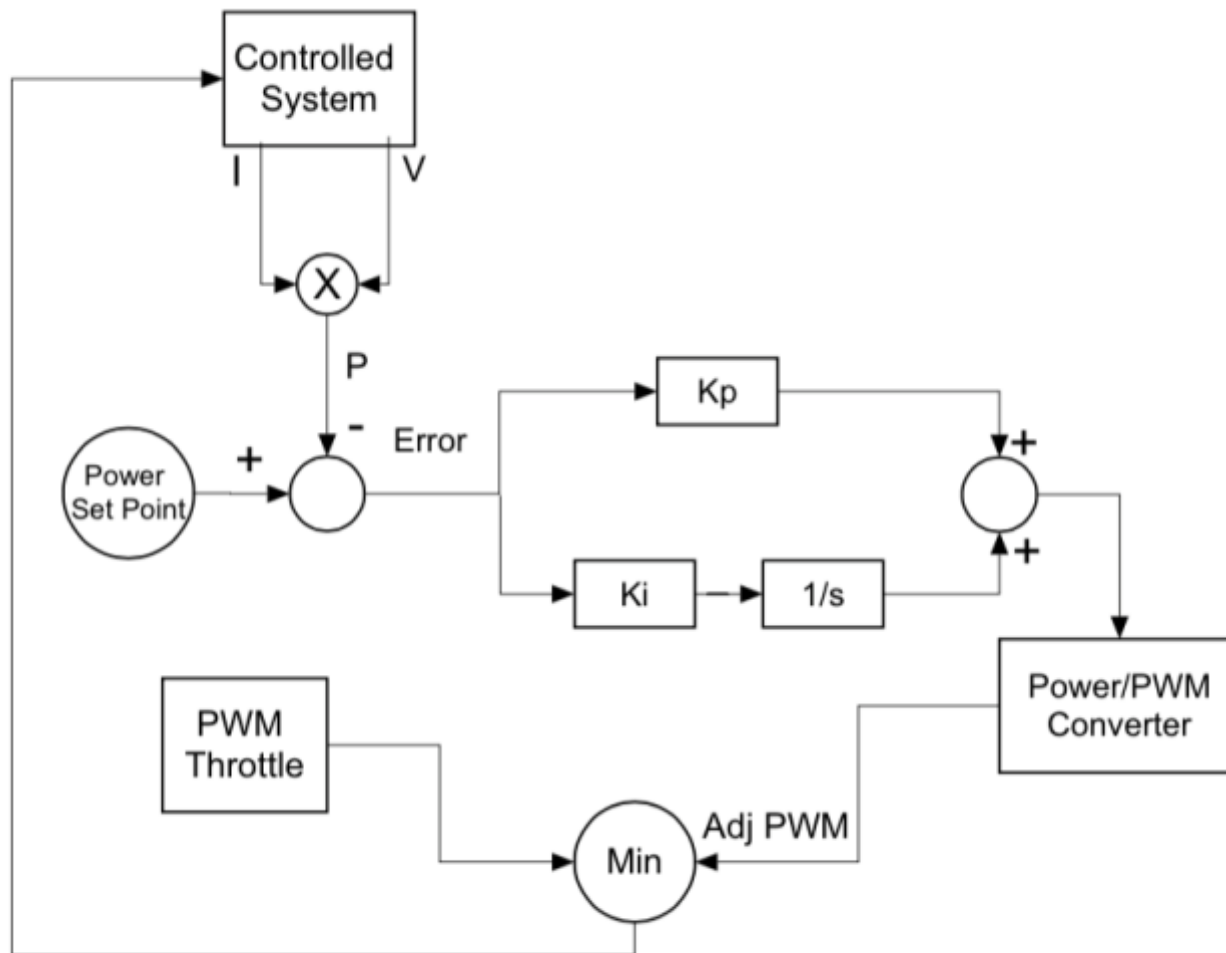
# 2015 EPS



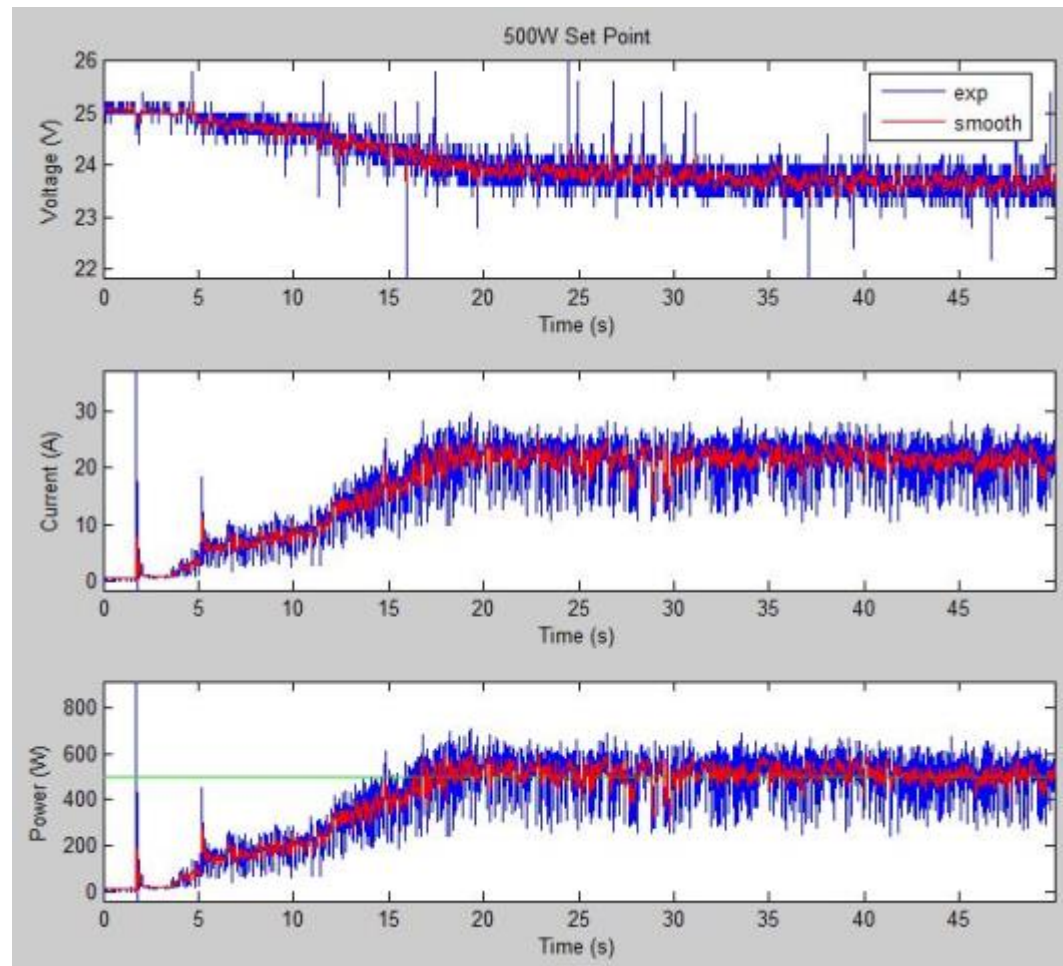
# What is a P.I Controller ?



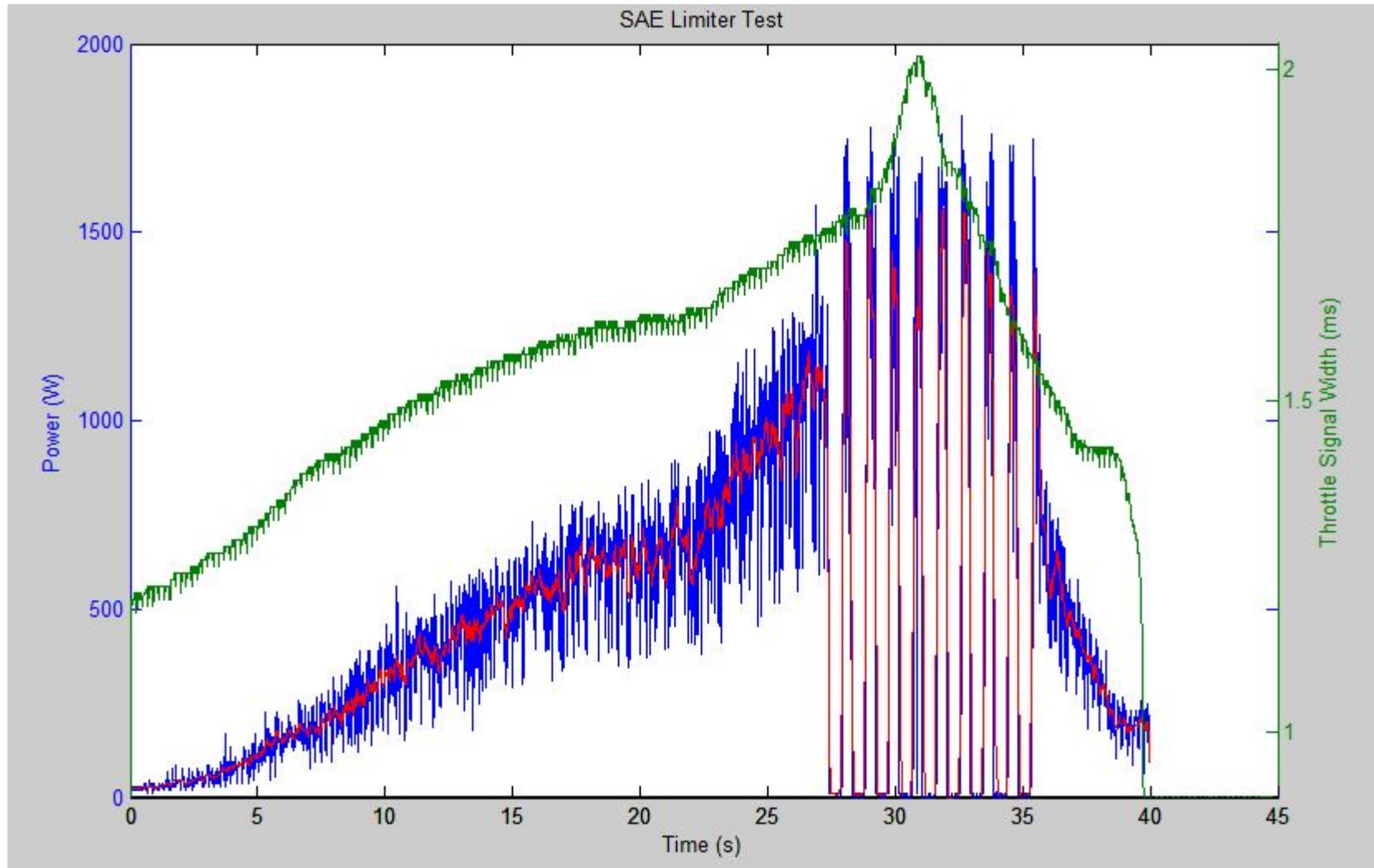
# P.I control Algorithm Block Diagram



# P.I Control Results

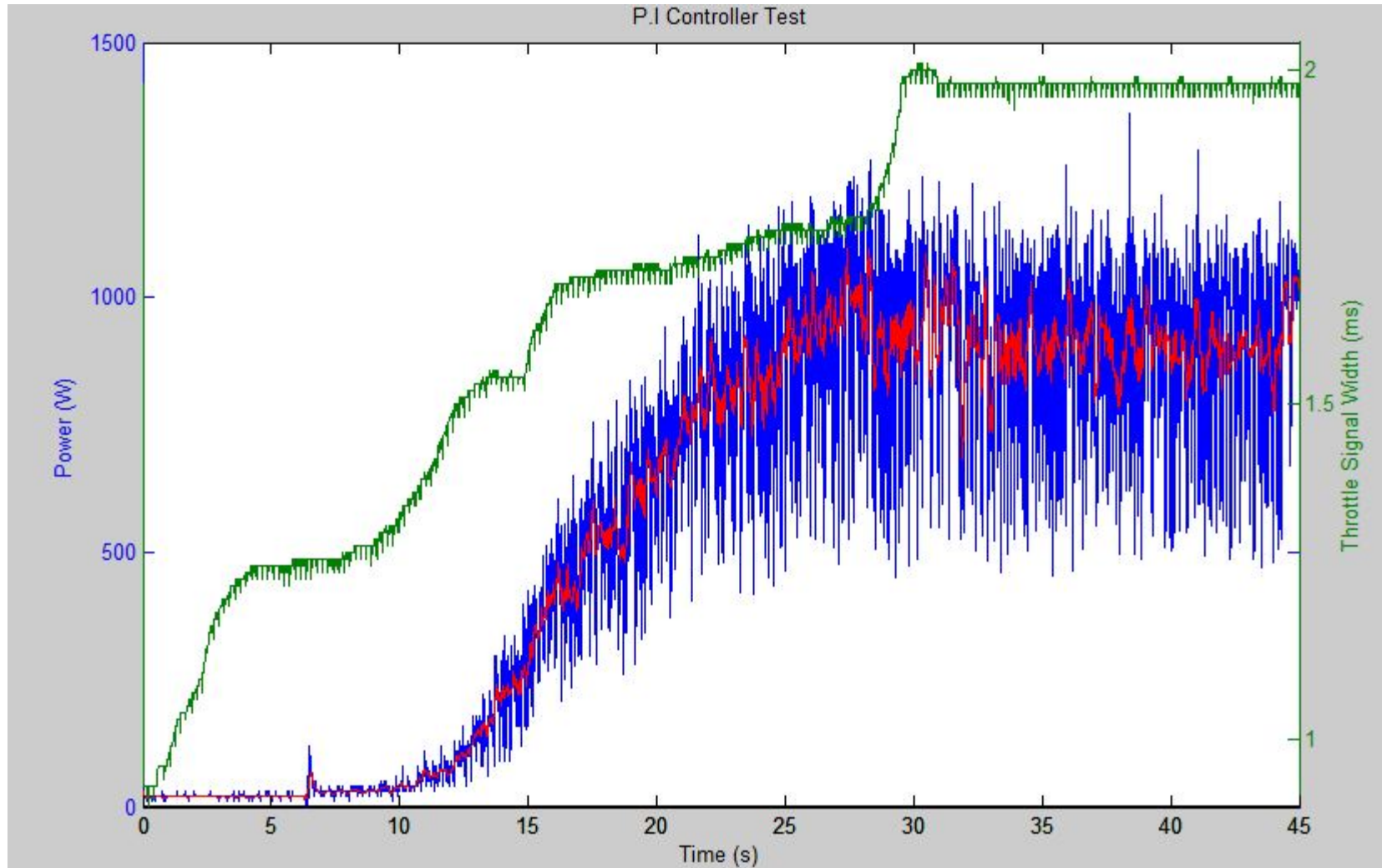


# Limiter Effect





# P.I Controller Effect

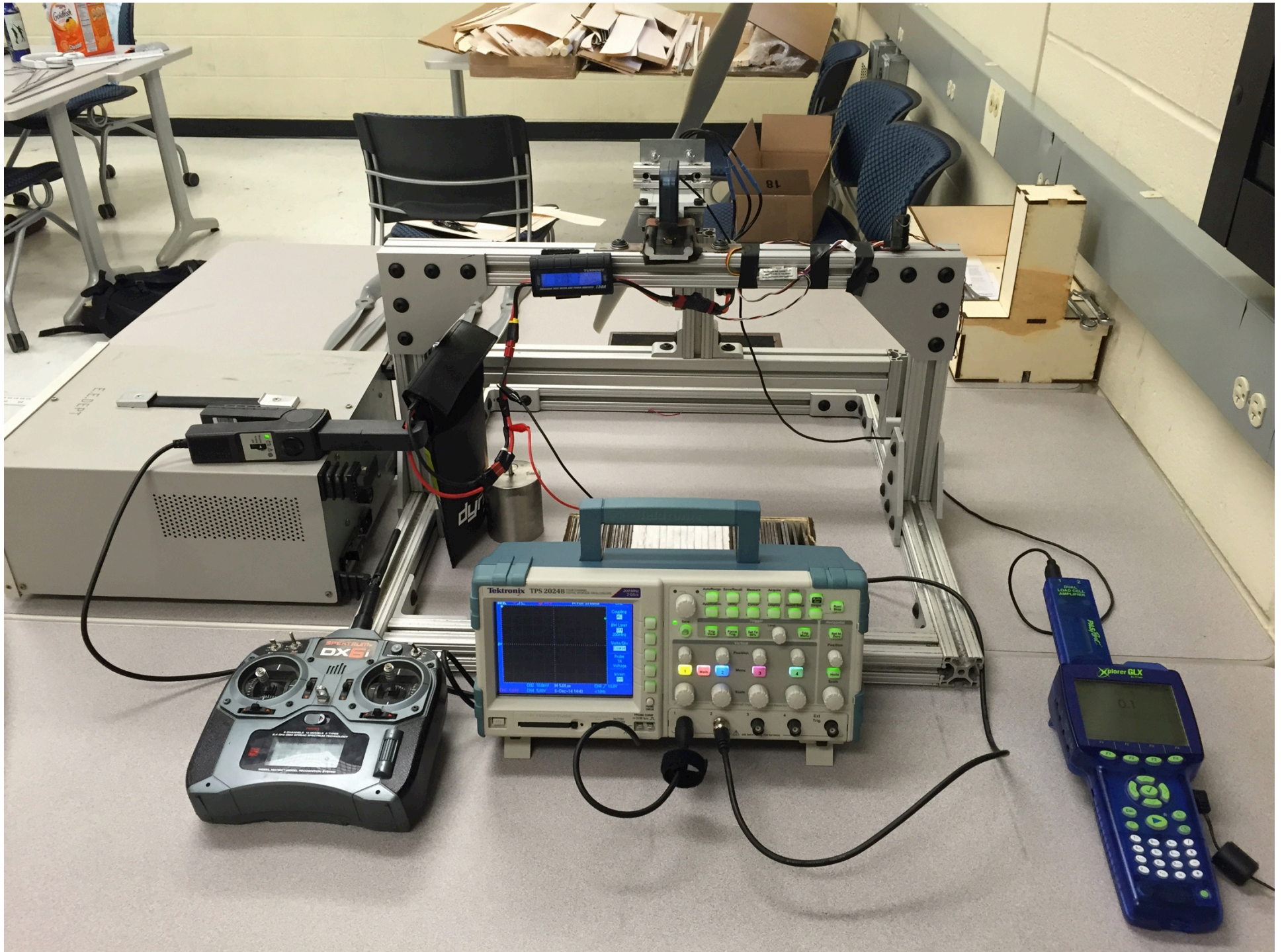


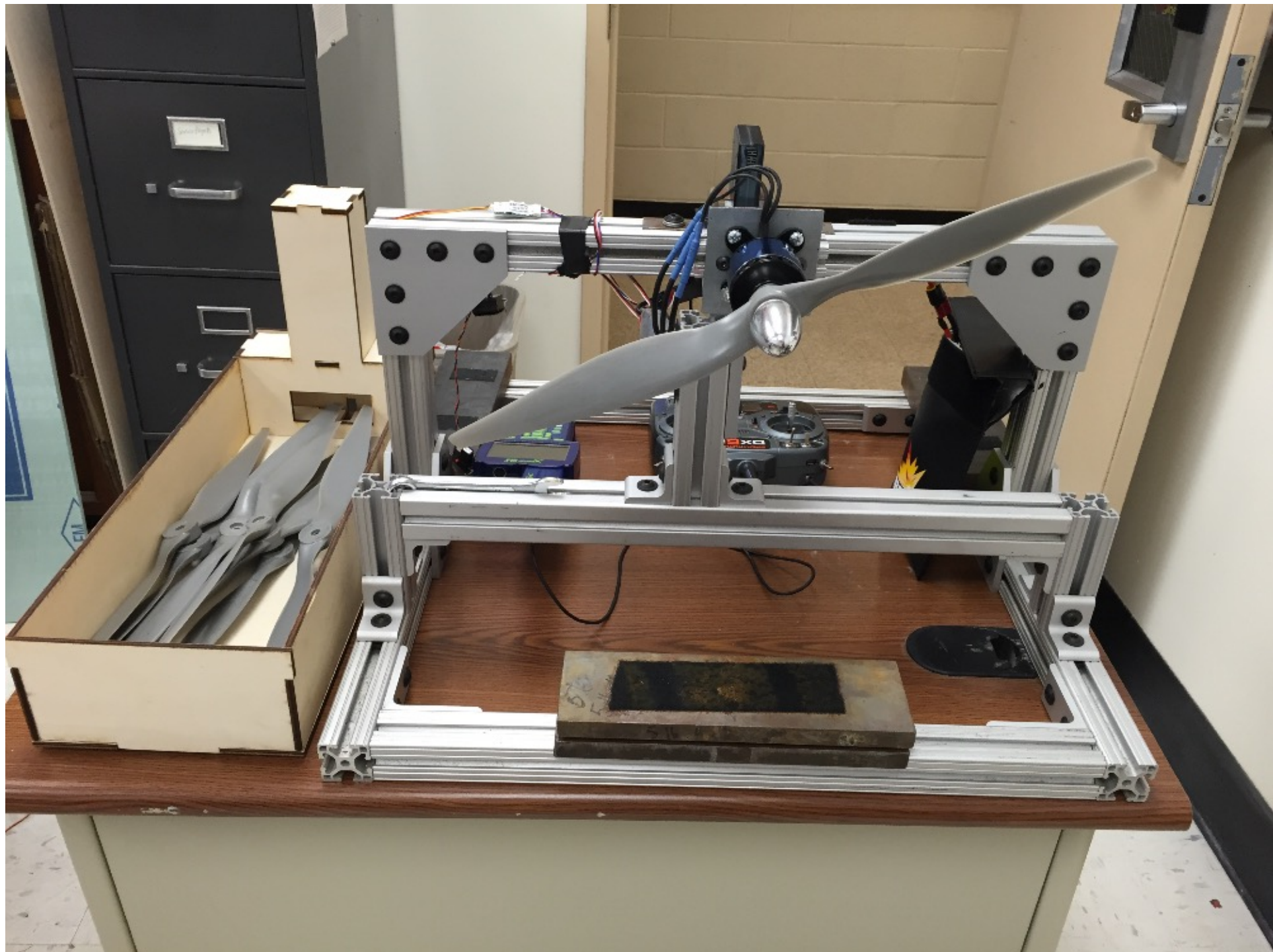
# Data Collection Tools

- Use of Thrust Test Bed
- Oscilloscope
- Current Probe
- BNC to alligator Clip connector
- Xplorer GLX Graphing Data Logger
- Dual Load Cell Amplifier

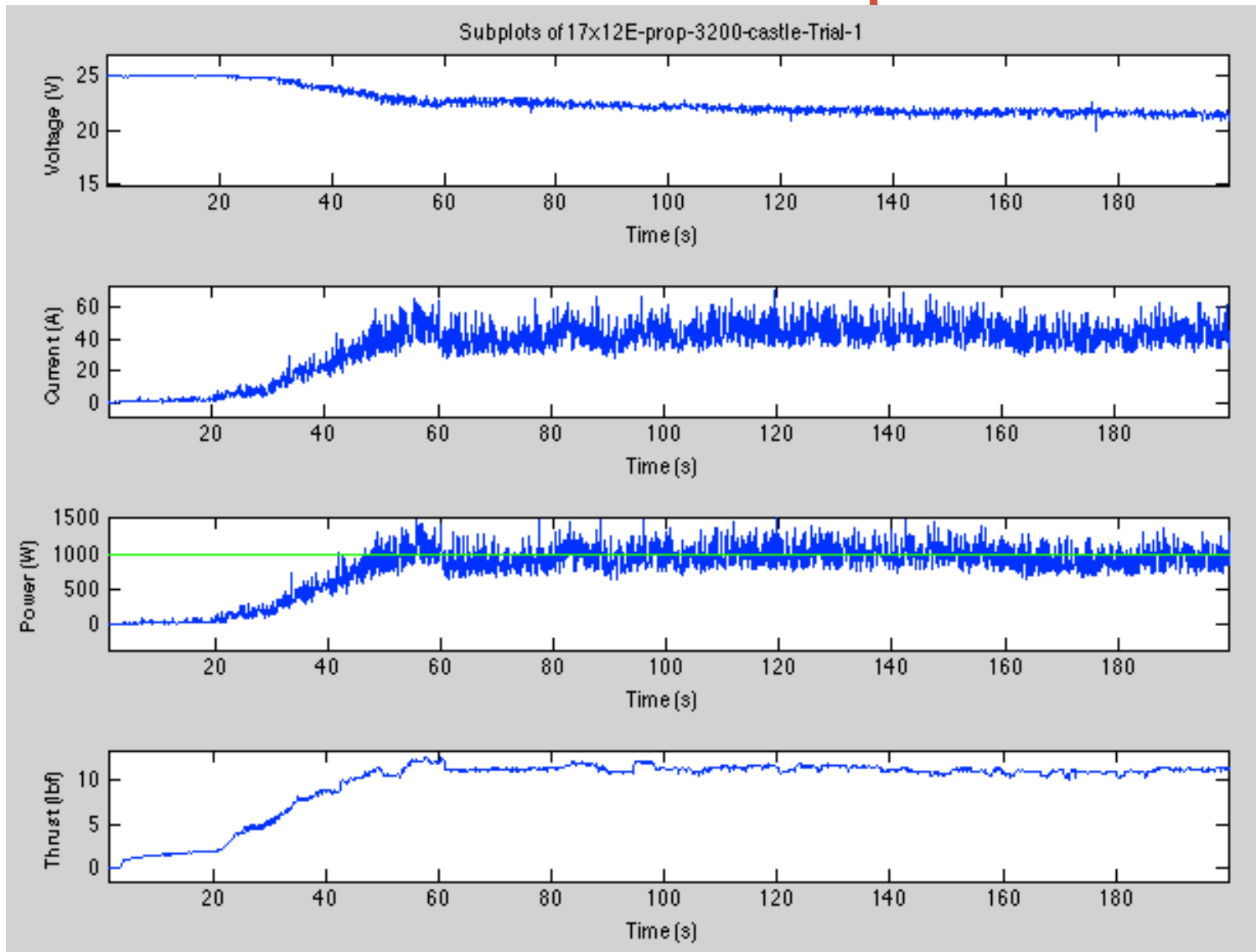
Propeller	Average Thrust (lbf)
17x12E	11.26
18x8	12.57
<b>18x8E</b>	<b>12.97</b>
18x10E	11.69
19x8	12.27
19x10	11.73
20x8E	12.11
20x10E	12.13



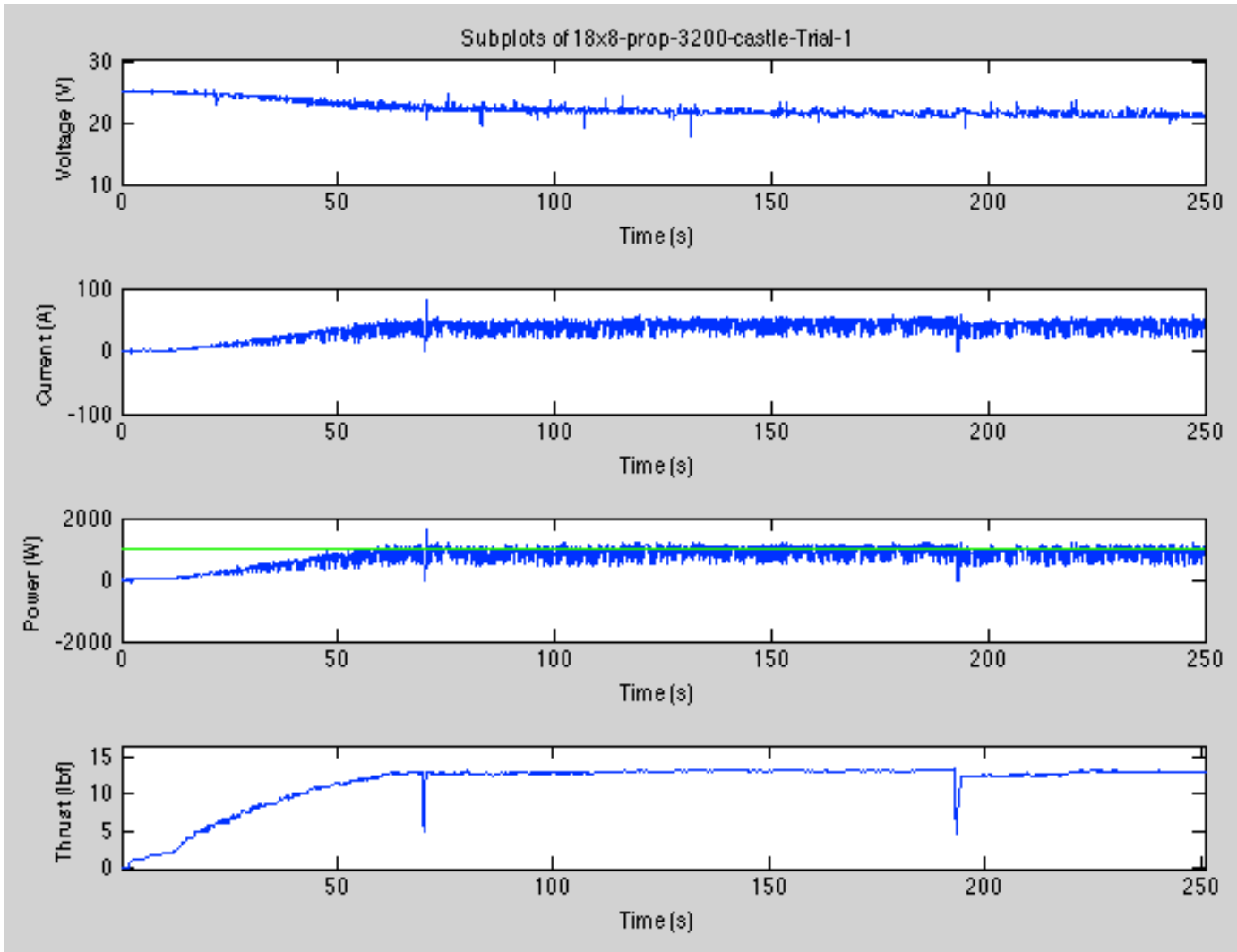




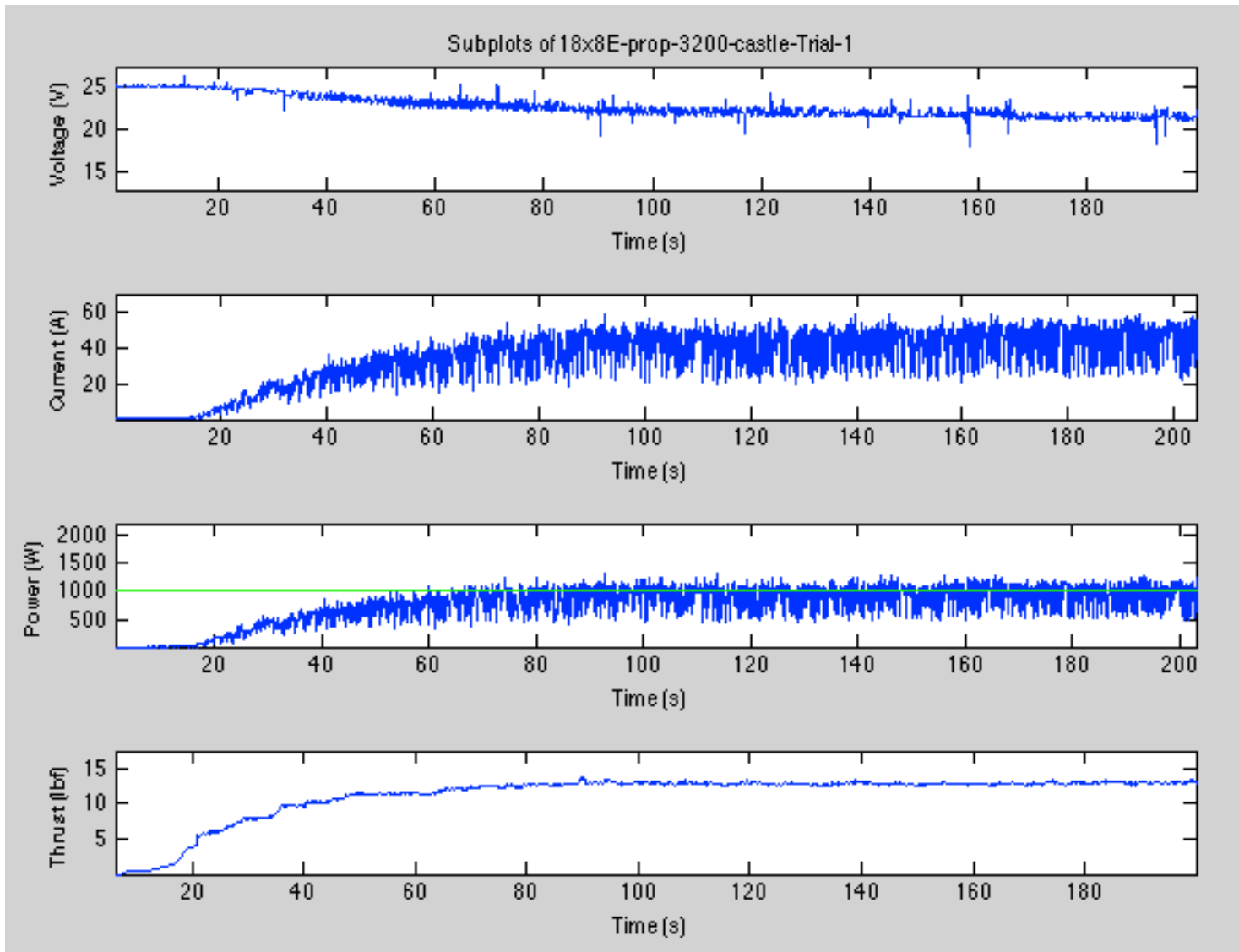
# 17x12E Prop



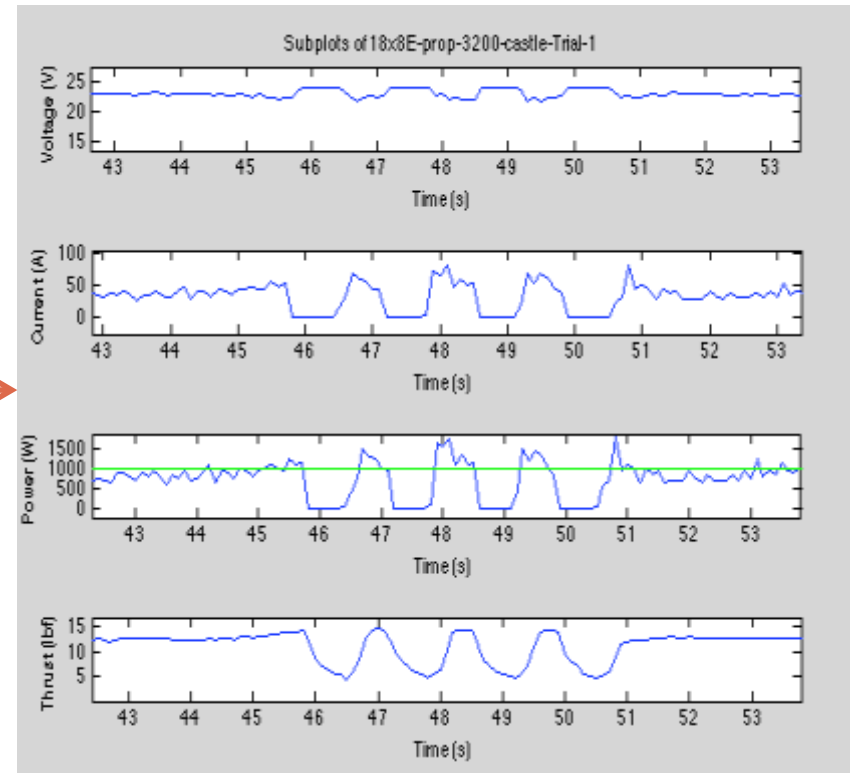
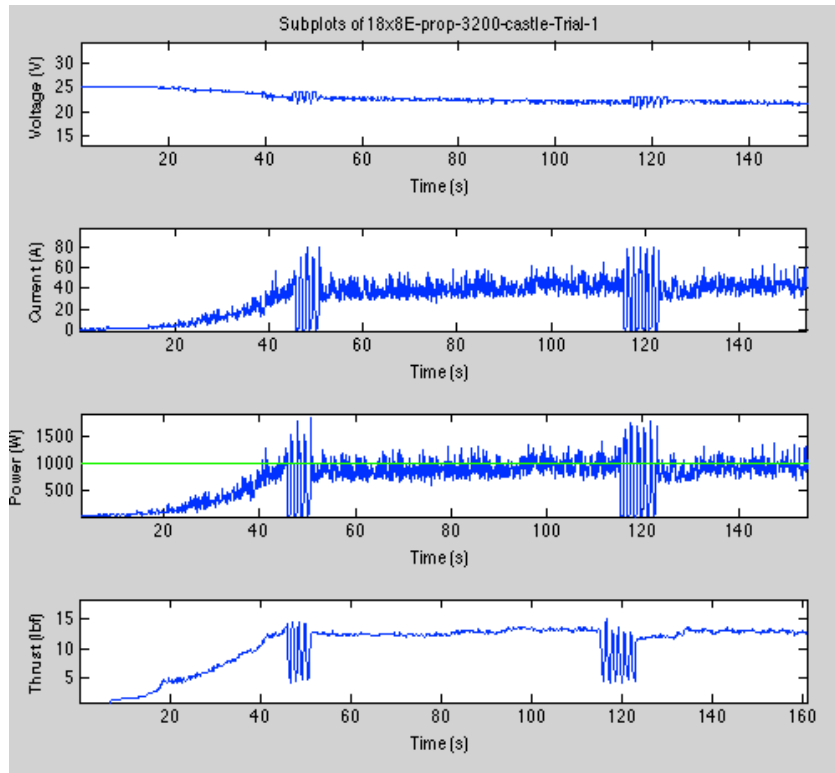
# 18x8 Prop



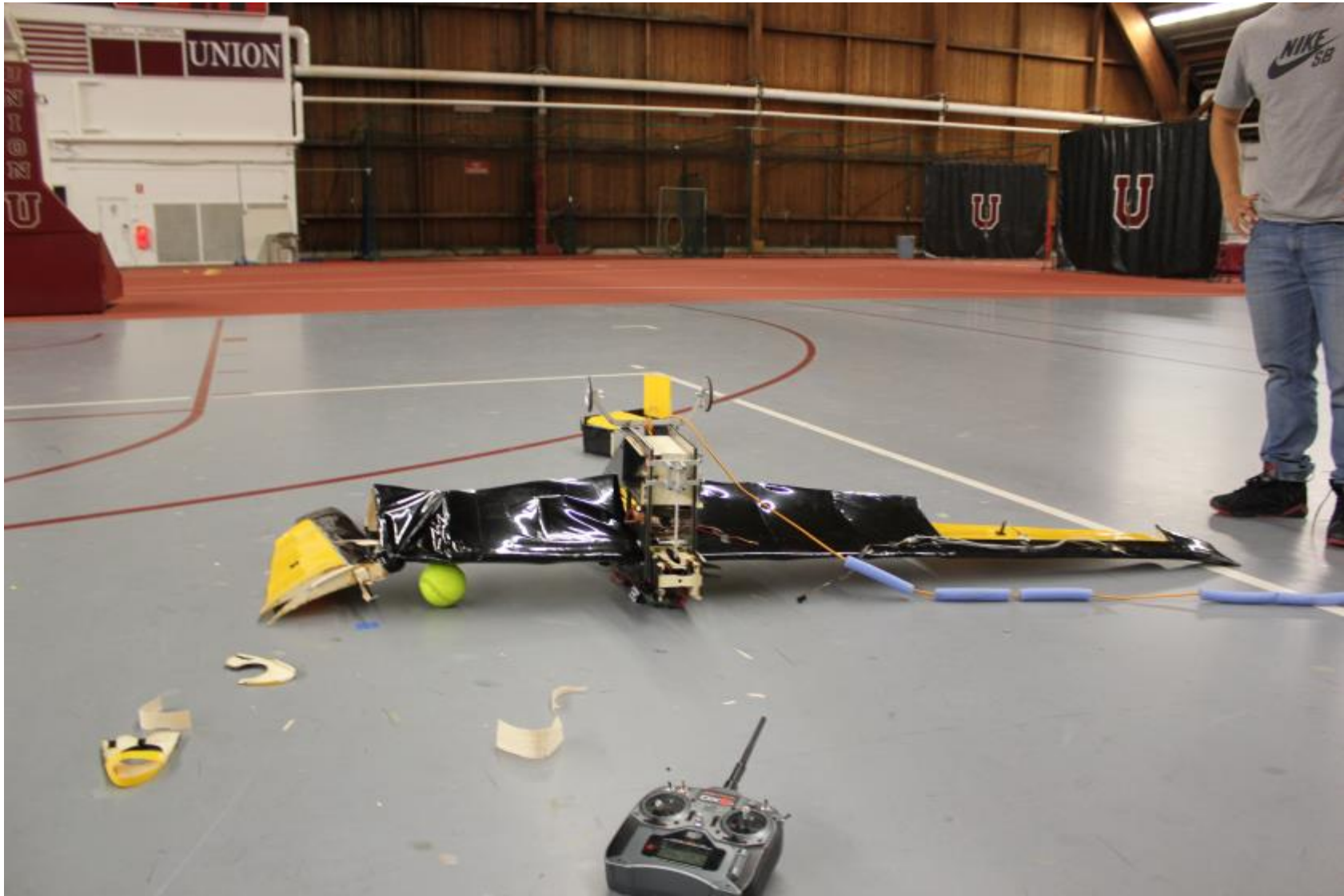
# 18x8E Prop



# Engaging Limiter







# Future Work

- Keep Testing P.I Controller
- Test more propellers with different pitches
- Use Ecalc as a reference for future testing
- Find lighter motor for future use
- Find a replacement EE



# Acknowledgments

- Professor John Spinelli
  - Professor James Hedrick
  - Professor Luke Dosiek
  - Professor Bradford Bruno
  - Gene Davison
  - Stan Gorski
  - Lisa Galeo
  - Rodolfo Santana
  - The 2014 Union College Aero Team
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# Questions?

